

FALCON SERVICE BULLETIN

FALCON 900

No 372-R1

JANUARY 26, 2015

MANDATORY

ATA 33-10

**LIGHTING
EXTERIOR LIGHTING
ELECTRICAL BONDING OF WHELEN FIN TIP ANTI-
COLLISION LIGHT**

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**LIGHTING
EXTERIOR LIGHTING
ELECTRICAL BONDING OF WHELEN FIN TIP ANTI-COLLISION LIGHT**

Initial issuance

August 22, 2007

List of effective pages :
1 to 18

Revision 1

January 26, 2015

List of effective pages :
1 to 19
List of modified pages :
1 to 11

REASON:The Service Bulletin is now rendered mandatory.

This revision is not applicable to aircraft already modified per the initial issuance of this Service Bulletin.

LIGHTING
EXTERIOR LIGHTING
ELECTRICAL BONDING OF WHELEN FIN TIP ANTI-COLLISION LIGHT

1. PLANNING INFORMATION

A. EFFECTIVITY

This Service Bulletin is applicable MYSTERE-FALCON 900 aircraft equipped with a WHELEN fin tip anti-collision light (without SATCOM fairing) per Service Bulletin F900-182 (M1682).

B. REASON

The investigation conducted on a Falcon 2000 aircraft that had been struck by lightning has shown that the electrical bonding of the fin tip anti-collision light may not be sufficient under certain circumstances, potentially resulting in damaging the aircraft wiring.

The purpose of this Service Bulletin is to improve the electrical bonding of the MYSTERE-FALCON 900 fin tip anti-collision light whose installation is similar to the F2000.

C. DESCRIPTION

After removing the fin tip fairing, the operation consists of the following:

- to install an electrical bonding strip in two parts between the anti-collision light and a rib of the fin,
- to modify the shield connection of the wiring,
- to reroute the wiring as required in accordance with the aircraft's serial number.

D. COMPLIANCE

Mandatory:

- This Service Bulletin is rendered mandatory by EASA Airworthiness Directive No. 2015-0006.
- As prescribed by the Airworthiness Authorities of the country of registration of the aircraft.

E. APPROVAL

This Service Bulletin covers DASSAULT AVIATION modification MF900 M5381 that has been approved under the authority of DOA nr. EASA.21J.051.

The technical content of this document is approved under the authority of DOA nr. EASA.21J.051.

F. LABOR

Estimated labor-hours: Refer to Service Bulletin Commercial Summary.

NOTE: These labor-hours only concern the work described in this Service Bulletin and do not include other maintenance work that may be performed on this occasion.

G. MATERIAL - PRICE AND AVAILABILITY

The modification kit may be obtained from either address listed below:

Western hemisphere: DASSAULT FALCON JET CORP.

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SPARES DISTRIBUTION CENTER
200 RISER ROAD
LITTLE FERRY, NJ 07643 U.S.A.

Telephone:

- CANADA and U.S.A.: 1-800-800-4036
- MEXICO: 001-800-800-4036
- Other countries: 1-201-541-4809

Fax:

- CANADA and U.S.A.: 1-800-800-4817
- MEXICO: 001-800-800-4817
- Other countries: 1-201-440-7021

Other continents:

DASSAULT AVIATION
Falcon Spares
BOITE POSTALE N°101
AEROPORT DU BOURGET
93350 - LE BOURGET Cedex (FRANCE)
Please contact your Dassault Aviation
Account Representative
Telephone: 33 (0)1.48.35.56.78
Fax: 33 (0)1.48.35.56.00
E/H dafsorder@dassault-aviation.com

Price and availability on request.

H. TOOLING - PRICE AND AVAILABILITY

Normal maintenance tooling.

I. WEIGHT AND BALANCE

Change in weight: None.

Change in balance with respect to 25% MAC: None.

J. REFERENCES

Aircraft Maintenance Manual:

- GENERAL MAINTENANCE AND SAFETY PRECAUTIONS (Refer to [TASK 20-00-00-910-801](#))
- PRINCIPLES AND CHECK OF THE ELECTRICAL BONDING (Refer to [TASK 20-20-00-910-801](#))
- OPERATIONAL TEST OF THE EXTERIOR LIGHTING (Refer to [TASK 33-40-00-710-801](#))

Illustrated Parts Catalog:

- ANTI-COLLISION LIGHTS (Refer to [IPC fig. 33-40-10-50](#))
- VERTICAL STABILIZER DOORS,RIB05 TO RIB10C (Refer to [IPC fig. 55-30-20-30](#))
- VERTICAL STABILIZER FIN TIP (Refer to [IPC fig. 55-30-40-10](#))

Service Bulletin F900-182 "Lights - Exterior lights - Replacement of fin anticollision light"

K. OTHER PUBLICATIONS AFFECTED

The following publications will be revised at a later date to account for the changes introduced by this Service Bulletin:

- Illustrated Parts Catalog.
- Maintenance Manual.
- Wiring Manual.

2. ACCOMPLISHMENT INSTRUCTIONS

A. PRELIMINARY STEPS

- (1) The aircraft must be in the maintenance configuration (Refer to **TASK 20-00-00-910-801**).
- (2) Remove the screws attaching upper fin tip fairing (**322A**) (Refer to **IPC fig. 55-30-40-10** , item 10).
- (3) Disengage upper fin tip fairing (**322A**), disconnect connector (322P) and remove upper fin tip fairing (**322A**).
- (4) On aircraft with Service Bulletin F900-182 applied, remove door (323FL) (Refer to **IPC fig. 55-30-20-30** , item 250).
- (5) Removal of the light unit assembly (Refer to **IPC fig. 33-40-10-50**):
 - Remove the compound bead coating the retainer (item 342).
 - Remove the retainer, glass cover and seal (items 342, 345 and 346).
 - Remove the light unit assembly (screws item 130).

NOTE: Remove and discard gasket (item 349) if installed.

B. PREPARING FAIRINGS F50B319, (A1), (A2) AND FGFB319, (A1)

NOTE: If not concerned, go to **step 2.C**.

- (1) Remove the contacts from connector (4LX).
- (2) Extract the wiring from the feedthrough at the upper part of rib 3.

NOTE: The wiring will be re-routed in accordance with the instructions of paragraph 2.F.

- (3) Machine the cutout in rib 3 in accordance with **fig. 8**.

C. ANTI-COLLISION LIGHT ELECTRICAL BONDING

In accordance with **fig. 1** or **fig. 2** depending on the P/N of the existing fairing:

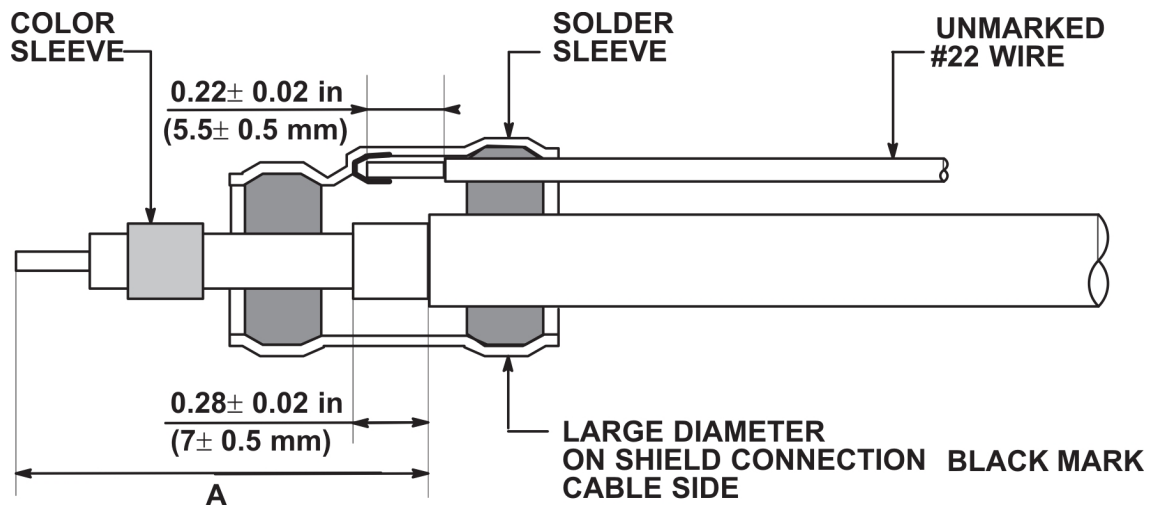
- (1) Cut out the run outs of the compound seal to obtain a plane mounting surface.
- (2) Set the mounting-base bonding strip (item 1), equipped with shim (item 8), into place on the fairing. Attach with two screws.
- (3) Fold the tab of the strip to a 90° angle with respect to the fairing's rib.
- (4) Pin the strip assembly (items 2 and 3) to the fairing.
- (5) Position the strip assembly (items 2 and 3) in contact against the RH side panel of the fairing and on strip (item 1). If necessary, rework the shape of the strips to obtain proper contact.
- (6) Remove the strip assembly (items 2 and 3).
- (7) Install double-sided adhesive tape on the face coming in contact with bonding strip (item 1).
- (8) Re-install strip assembly (items 2 and 3). Pin to hold in contact against the fairing.
- (9) Install strip (item 1) over the strip assembly, pressing firmly at the contact area).
- (10) Remove the strip assembly (items 1, 2 and 3).
- (11) Counterdrill bonding strip (item 2) to a diameter of 0.1 in (2.5 mm). Disconnect the strips and discard the adhesive tape, remove all remaining adhesive.

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- (12) Enlarge the holes drilled in the strips to a diameter of 0.126 in (3.2 mm). Assemble bonding strips (items 1 and 2) using rivets (item 4). Coat the faces of the strips with **covering sealant** making a bead at the contour edge with **bead sealant**.
- (13) Clean the contact surfaces of the light unit assembly, of the strips, and of the fairing using **MEK**.
- (14) Coat the strip assembly (items 2 and 3) with **interlaying sealant**.
- (15) Set the strip assembly and the light unit assembly into place in the fairing without making electrical connections and attach the light assembly to the fairing. Smooth out the PR in excess around the strips (items 2 and 3).
- (16) Use a shim to wedge strip assembly (items 2 and 3) in contact against the fairing's wall. Allow to cure.
- (17) Temporarily position the fairing at the fin tip and mark the location of strip (item 3).
- (18) Strip the paint off the fin tip rib at the location marked for strip (item 3). Protect the stripped area with **alodine**.

D. MODIFYING THE WIRING

Modify the shield connection in accordance with instructions depicted hereafter and using **fig. 3**, **fig. 4** and **fig. 5** in accordance lights' installation on the aircraft.


Dimension A:

- Connector with front extraction: 1.38 in (35 mm).
- Connector with rear extraction: 1.97 in (50 mm).

Self-soldering sleeves must be staggered at the extremity of the wiring assemblies so as to prevent a bulge on the bundle.

INSTALLING SELF-SOLDERING SLEEVES

- After stripping the shielded cable and the ground lead wire, unwind the shielding braid of cable **without overlapping the shielding strands**.
- Position the self-soldering sleeve on the stripped part of the cable.

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- Insert the stripped part of the ground lead wire on the inside of the solder preform. Make sure that the ground lead wire is installed against the guide provided for this purpose.
- With hot air or infrared generator equipped with an appropriate reflector, start heating the small diameter of the self-soldering sleeve, then move up to the solder preform until it melts and flows. Complete the operation on the large side of the self-soldering sleeve. It is recommended to position the solder preform at a 90° angle maximum in relation to the heat source. The solder must always be visible to the operator.
- The insulating sheath of the shielded cable must not be blistered nor burnt.

E. RE-ROUTING THE WIRING WITH WIRING ROUTED OVER RIB 2

In accordance with **fig. 6**:

- (1) Remove the WHELEN light unit assembly.
- (2) Drill a 0.19 in (4.9 mm) diameter hole through the rib of the fairing.
- (3) Install new clamping stud (item 5) on rib 3.
- (4) Unclamp the wiring from rib 2.
- (5) Clamp the wiring to rib 3.
- (6) Connect connector (4LX) to the WHELEN light.
- (7) Attach the light unit assembly to the fairing.
- (8) Install the gasket, the glass cover, and the retainer (Refer to [IPC fig. 33-40-10-50](#) , items 346, 345, 342).
- (9) Apply **bead sealant** around the retainer.

F. RE-ROUTING THE WIRING WITH WIRING ROUTED AROUND THE BACK OF FAIRING

In accordance with **fig. 7** (Fairing P/N F50B319BX or FGFB319A2, (A3)) or **fig. 8** (Fairing P/N F50B319, (A1), (A2) or FGFB319, (A1))

- (1) Remove the WHELEN light unit assembly.
- (2) Drill a 0.19 in (4.9 mm) diameter hole through the rib 3 of the fairing.
- (3) Install new clamping stud (item 5) on rib 3.
- (4) Remove contacts from connector (4LX).
- (5) Adjust the wiring length for proper re-routing in accordance with **fig. 7**.
- (6) Fit the cut wires with new contacts and insert them into connector (4LX).
- (7) Clamp the wiring to rib 3.
- (8) Connect connector (4LX) to the WHELEN light.
- (9) Attach the light unit assembly to the fairing.
- (10) Install the gasket, the glass cover, and the retainer (Refer to [IPC fig. 33-40-10-50](#) , items 346, 345, 342).
- (11) Apply **bead sealant** around the retainer.

G. FINAL STEPS

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- (1) Use a permanent ink pen to record the number of this Service Bulletin on "Service Bulletin Compliance" label (7-fig. 7).
- (2) Use **interlaying sealant** to bond the "Service Bulletin Compliance" label to the inner skin of the fairing next to the P/N (7-fig. 7).
- (3) Place the fairing on the fin tip, connect connector (322P) and attach the fairing.
- (4) Re-install the door (323FL) (Refer to **IPC fig. 55-30-20-30** , item 250) (aircraft with Service Bulletin F900-182 applied).
- (5) Measure the electrical bonding across the anti-collision light base plate and the reference point located in the mechanic's servicing compartment (Refer to **TASK 20-20-00-910-801**).
The measured value must not exceed 2 mΩ.
- (6) Perform the operational test of the upper anti-collision light (Refer to **TASK 33-40-00-710-801**).

H. RECORDING

Record compliance with this Service Bulletin in the appropriate aircraft documents.

Fill out the card located at the end of the Service Bulletin and mail or fax it to Dassault Aviation or Dassault Falcon Jet, or fill out the electronic Service Bulletin reply form located in the "Service Bulletin" page on the Falcon portal.

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3. MATERIAL INFORMATION
A. MODIFICATION KIT
KIT P/N SBF900EX0285A

APPLICABLE TO AIRCRAFT EQUIPPED WITH FAIRINGS F50B319BX OR FGFB319A2, (A3)

KIT P/N SBF900EX0285A - Mechanical kit				
NEW P/N	QTY	KEY WORD	INSTRUCTIONS DISPOSITION	ITEM FIG
FGFB319100010	1	Mounting-base bonding strip	Supply already assembled	- (1-fig. 1) -
FGFB319100038	1	Shim		- (8-fig. 1) -
FGFB319100036	1	Strip	Supply already assembled	- (2-fig. 1) -
FGFB319100037A1	1	Strip		- (3-fig. 1) -
NAS1919M04S02WU	3	Rivet		- (4-fig. 1) -
78753	1	Studded support		- (5-fig. 6) - (5-fig. 7) -
78750-02	1	Clamp		- (6-fig. 6) - (6-fig. 7) -
79160	1	"Service Bulletin Compliance" label		- (7-fig. 6) - (7-fig. 7) -

KIT P/N SBF900EX0285A - Electrical kit				
NEW P/N	QTY	KEY WORD	INSTRUCTIONS DISPOSITION	FIG
01EQTC0S20-05-119-18	1	Lug	Ground 322J	- fig. 3 - fig. 4 - fig. 5
01C0K302-16S01	1	Contact	Ground 322J	- fig. 3 - fig. 4 - fig. 5
01CAN210-1-20	1	#20 gauge wire L= 3.3 ft (1 m)	Ground 322J	- fig. 3 - fig. 4 - fig. 5
01C0A54143FC16	4	Blanking pin	322J/P	- fig. 3 - fig. 4 - fig. 5
01EQTRB23-2R	1	Self-soldering sleeve	4LX	- fig. 3 - fig. 4 - fig. 5
01CAN210-1-22	1	#22 gauge CF wire L = 1 ft (0.3 m) See NOTE 1	4LX	- fig. 3 - fig. 4 - fig. 5
01CAN203-1-22Y	1	#22 gauge KZ wire L = 1 ft (0.3 m)		

DASSAULT AVIATION Proprietary Data

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KIT P/N SBF900EX0285A - Electrical kit				
NEW P/N	QTY	KEY WORD	INSTRUCTIONS DISPOSITION	FIG
		See NOTE 2		
01EQTC0S20A03-071-22	1	Lug	4LX	- fig. 3 - fig. 4 - fig. 5
01C0K345-16S01	2	Contact	4LX	- fig. 3 - fig. 4 - fig. 5
01EQTEC1-1	1	Insulating sleeve	322P	- fig. 3 - fig. 4 - fig. 5
01EQTEC1-1	4	Insulating sleeve - See NOTE 3	322J/P	- fig. 4 -
01EQTPRL08B01	1	Splice - See NOTE 3	SP3LX-A	- fig. 4 -
GND322J-B	2	Black sleeve with white marking 01PR0TMS3-32BK	322J	- fig. 3 - fig. 4 - fig. 5
GND312J-B	2	Black sleeve with white marking 01PR0TMS3-32BK	312J	- fig. 3 - fig. 4 - fig. 5
A	1	Black sleeve with white marking 01PR0TMS3-32BK	4LX	- fig. 3 - fig. 4 - fig. 5
B	1	Black sleeve with white marking 01PR0TMS3-32BK	4LX	- fig. 3 - fig. 4 - fig. 5
78750-01	5	Clamp	4LX	- fig. 3 - fig. 4 - fig. 5

NOTE 1: Use on MYSTERE-FALCON 900.

NOTE 2: Do not use on MYSTERE-FALCON 900.

NOTE 3: Use on MYSTERE-FALCON 900 with SB F900-182 applied.

KIT P/N SBF50-0481A

APPLICABLE TO THOSE AIRCRAFT EQUIPPED WITH FAIRINGS F50B319, (A1), (A2) OR FGFB319, (A1)

KIT P/N SBF50-0481A - Mechanical kit				
NEW P/N	QTY	KEY WORD	INSTRUCTIONS DISPOSITION	ITEM FIG
FGFB319100010W1	1	Mounting-base bonding strip	Supply already assembled	- (1-fig. 2) -
FGFB319100038	1	Shim		- (8-fig. 2) -

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KIT P/N SBF50-0481A - Mechanical kit				
NEW P/N	Q T Y	KEY WORD	INSTRUCTIONS DISPOSITION	ITEM FIG
FGFB319100036W1	1	Strip	Supply already assembled	- (2-fig. 2) -
FGFB319100037W2	1	Strip		- (3-fig. 2) -
NAS1919M04S02WU	3	Rivet		- (4-fig. 2) -
78753	1	Studded support		- (5-fig. 8) -
78750-02	1	Clamp		- (6-fig. 8) -
79160	1	"Service Bulletin Compliance" label		- (7-fig. 6) - (7- fig. 7) - (7-fig. 8) -

KIT P/N SBF50-0481A - Electrical kit				
NEW P/N	Q T Y	KEY WORD	INSTRUCTIONS DISPOSITION	FIG
01EQTC0S20-05-119-18	1	Lug	Ground 322J	- fig. 3 - fig. 4 - fig. 5
01C0K302-16S01	1	Contact	Ground 322J	- fig. 3 - fig. 4 - fig. 5
01CAN210-1-20	1	#20 gauge wire L= 3.3 ft (1 m)	Ground 322J	- fig. 3 - fig. 4 - fig. 5
01C0A54143FC16	4	Blanking pin	322J/P	- fig. 3 - fig. 4 - fig. 5
01EQTRB23-2R	1	Self-soldering sleeve	4LX	- fig. 3 - fig. 4 - fig. 5
01CAN210-1-22	1	#22 gauge CF wire L = 1 ft (0.3 m) - See NOTE 1	4LX	- fig. 3 - fig. 4 - fig. 5
01CAN203-1-22Y	1	#22 gauge KZ wire L = 1 ft (0.3 m) - See NOTE 2		
01EQTC0S20A03-071-22	1	Lug	4LX	- fig. 3 - fig. 4 - fig. 5
01C0K345-16S01	2	Contact	4LX	- fig. 3 - fig. 4 - fig. 5
01EQTEC1-1	1	Insulating sleeve	322P	- fig. 3 - fig. 4 - fig. 5
01EQTEC1-1	4	Insulating sleeve - See NOTE 3	322J/P	- fig. 4 -
01EQTPRL08B01	1	Splice - See NOTE 3	SP3LX-A	- fig. 4 -

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KIT P/N SBF50-0481A - Electrical kit				
NEW P/N	QTY	KEY WORD	INSTRUCTIONS DISPOSITION	FIG
GND322J-B	2	Black sleeve with white marking 01PR0TMS3-32BK	322J	- fig. 3 - fig. 4 - fig. 5
GND312J-B	2	Black sleeve with white marking 01PR0TMS3-32BK	312J	- fig. 3 - fig. 4 - fig. 5
A	1	Black sleeve with white marking 01PR0TMS3-32BK	4LX	- fig. 3 - fig. 4 - fig. 5
B	1	Black sleeve with white marking 01PR0TMS3-32BK	4LX	- fig. 3 - fig. 4 - fig. 5
78750-01	5	Clamp	4LX	- fig. 3 - fig. 4 - fig. 5

B. LOCAL PROCUREMENT

As indicated in Consumable Products Manual (CPM):

- alodine.
- interlaying sealant.
- bead sealant.
- MEK or equivalent.
- covering sealant.
- Green CELOMER varnish 6060B.
- Locking wire.

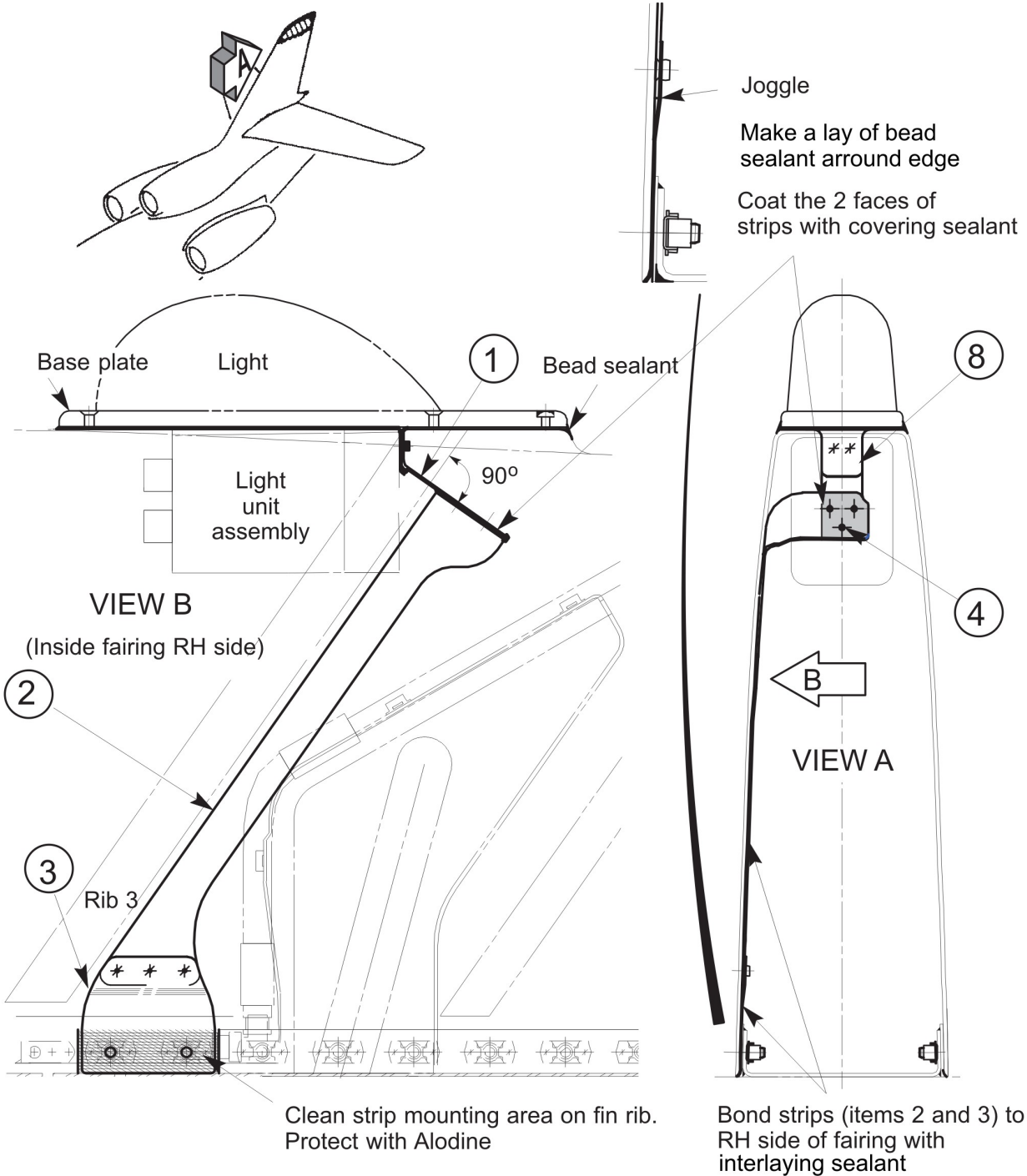


Figure 1: ELECTRICAL BONDING ON AIRCRAFT WITH FAIRINGS F50B319BX AND FGFB319A2 (A3)

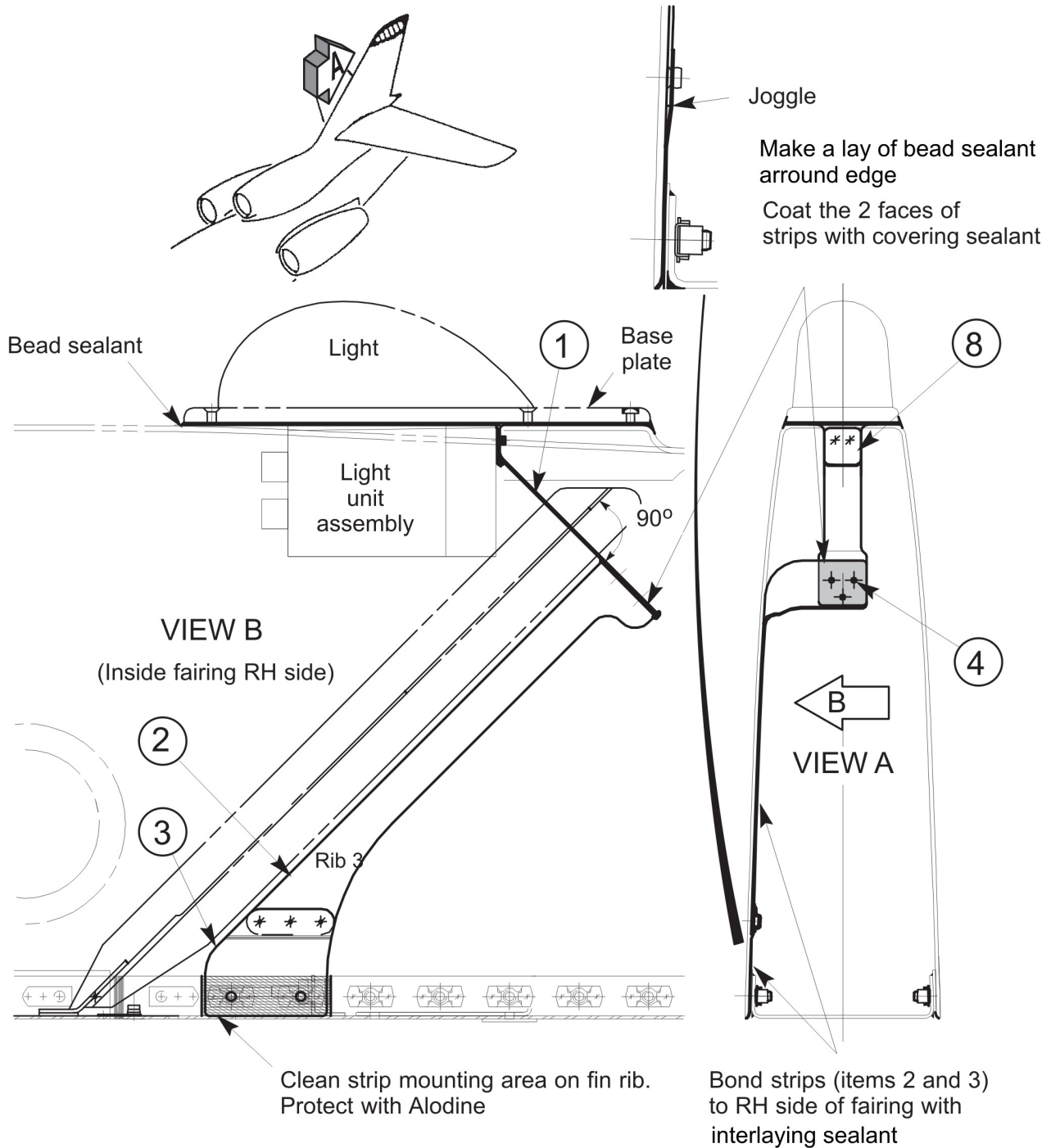


Figure 2: ELECTRICAL BONDING ON AIRCRAFT WITH FAIRINGS F50B319, (A1), (A2) AND FGFB319, (A1)

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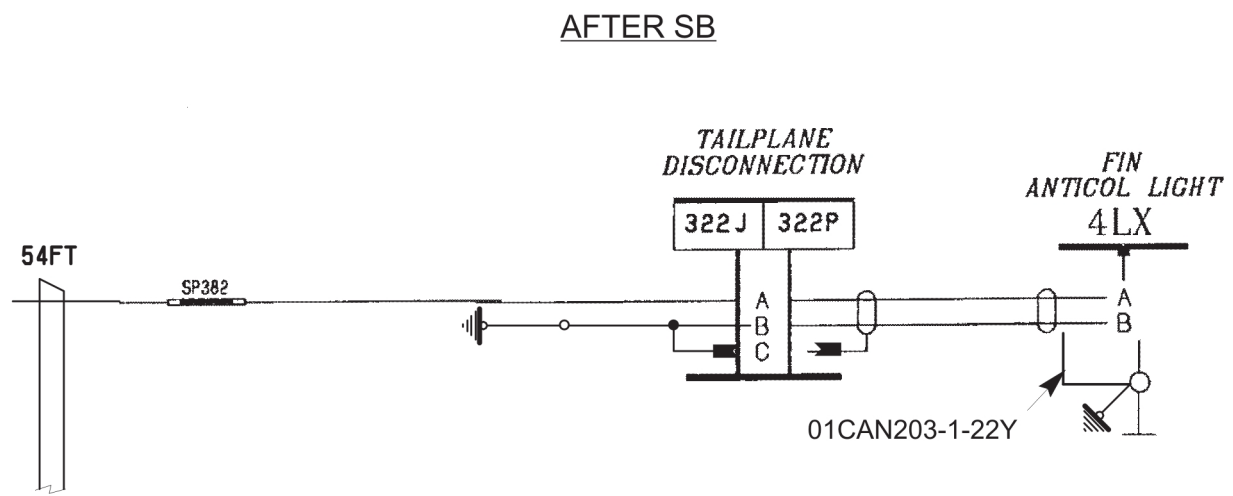
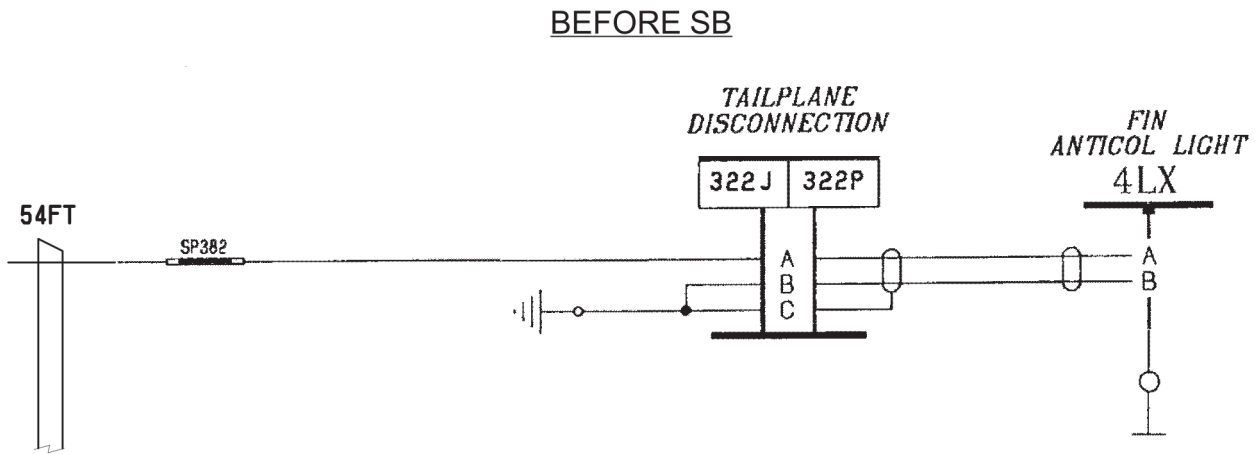
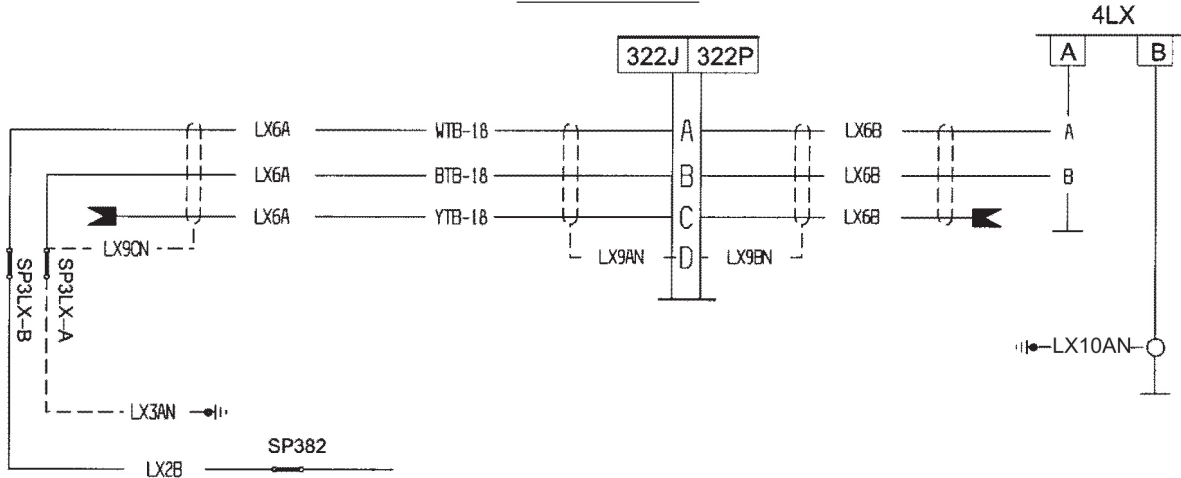


Figure 3: AIRCRAFT F900B WITH MODIFICATION MF900 M1682 APPLIED IN PRODUCTION LINE

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BEFORE SB



INCORPORATE INTO
WD 33-44-02

AFTER SB

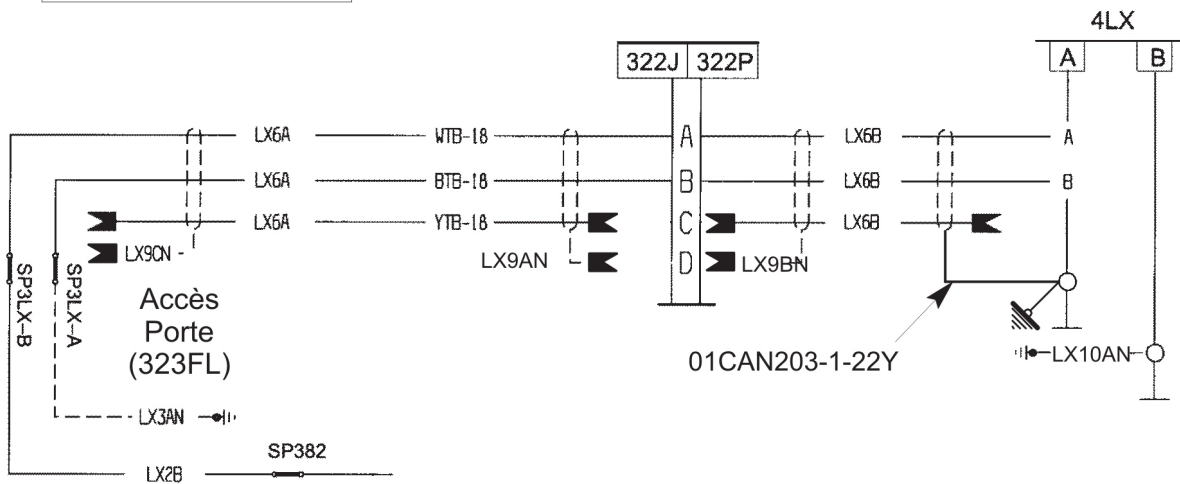
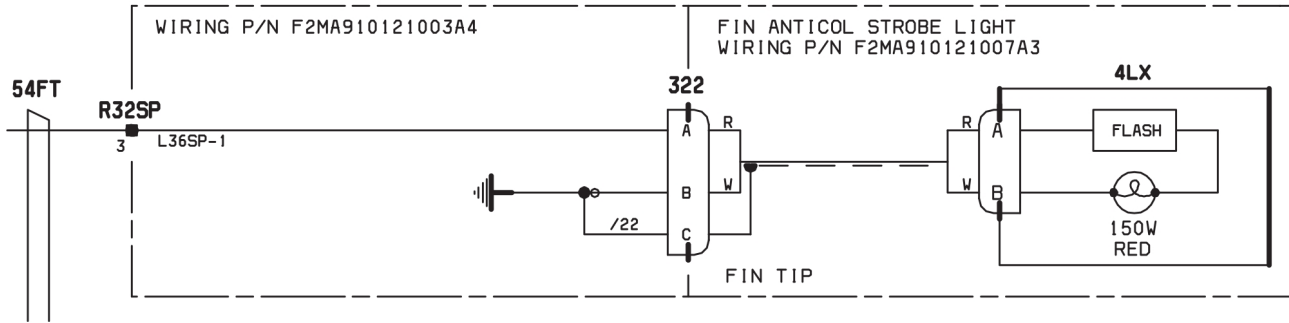


Figure 4: AIRCRAFT F900B WITH F900-182 APPLIED

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BEFORE SB



INCORPORATE INTO
WD 33-42-00

AFTER SB

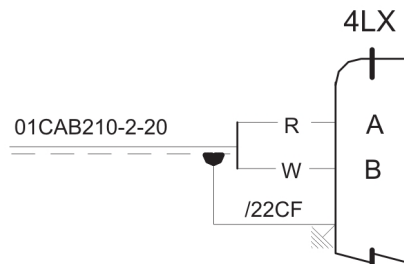
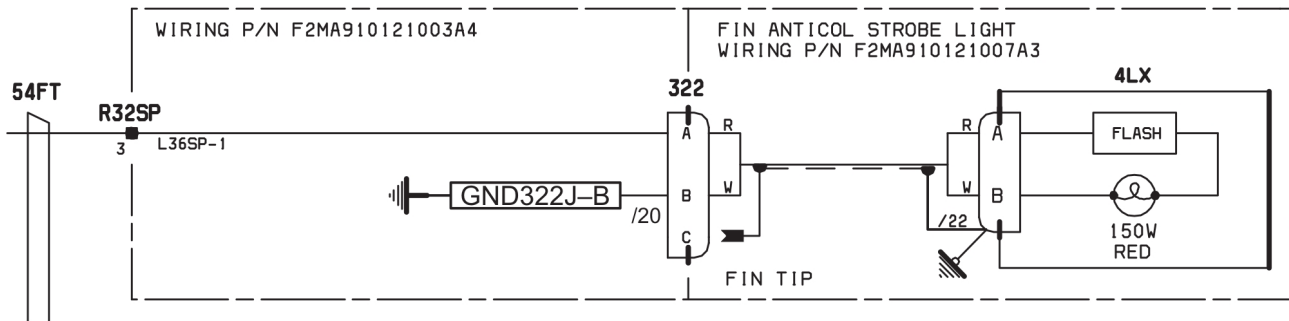
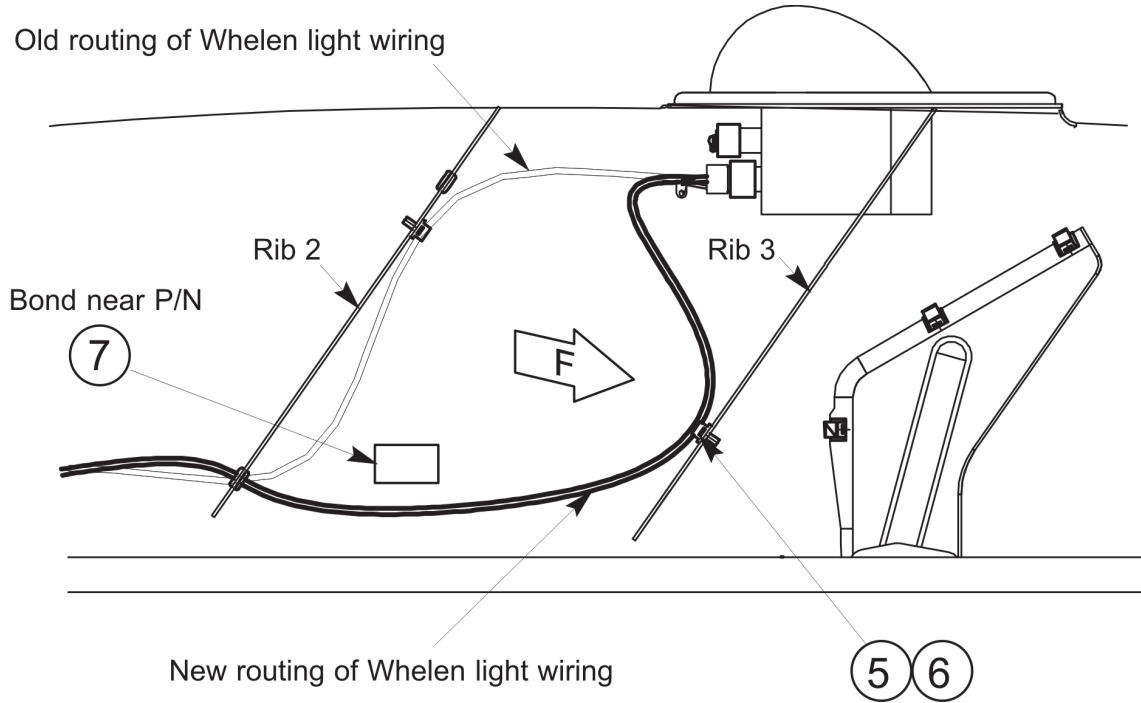


Figure 5: AIRCRAFT F900C S/N 179 AND SUBSEQUENT



Drill 0.19 in (4.9 mm) diameter hole for installation of stud (item 5)

View F

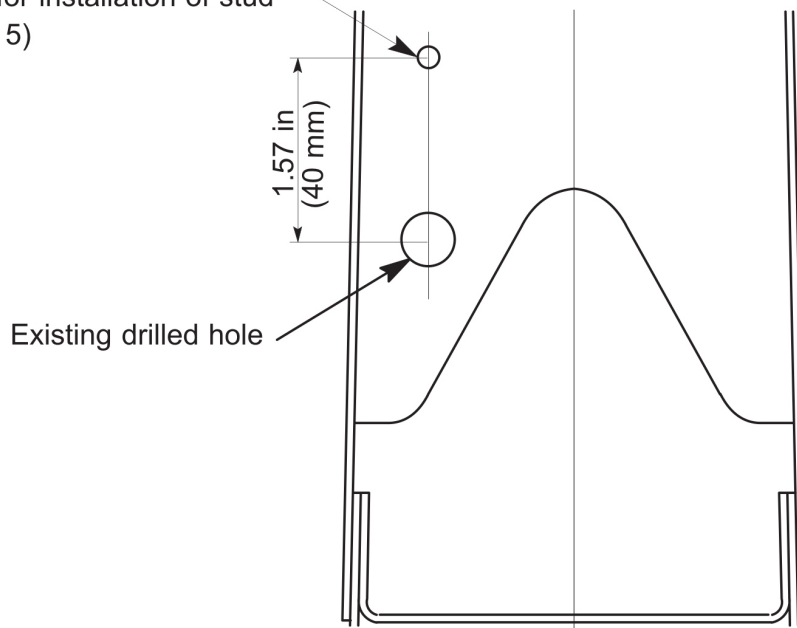
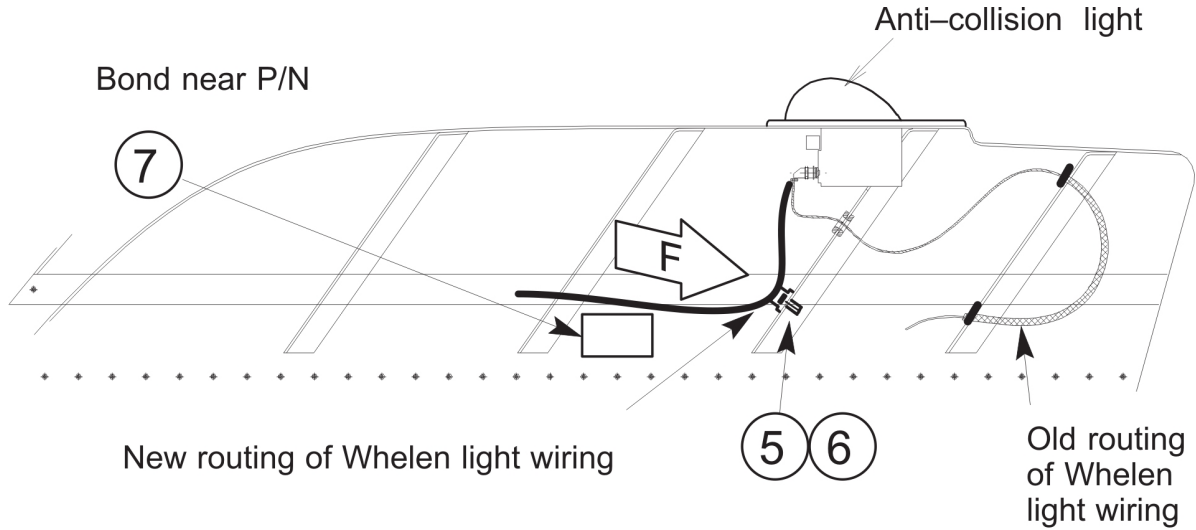


Figure 6: FAIRING P/N F50B319BX OR FGFB319A2 (A3) - RE-ROUTING THE WIRING - AIRCRAFT WITH WIRING ROUTED OVER RIB 2



Drill 0.19 in (4.9 mm) diameter hole for installation of stud (item 5)

View F

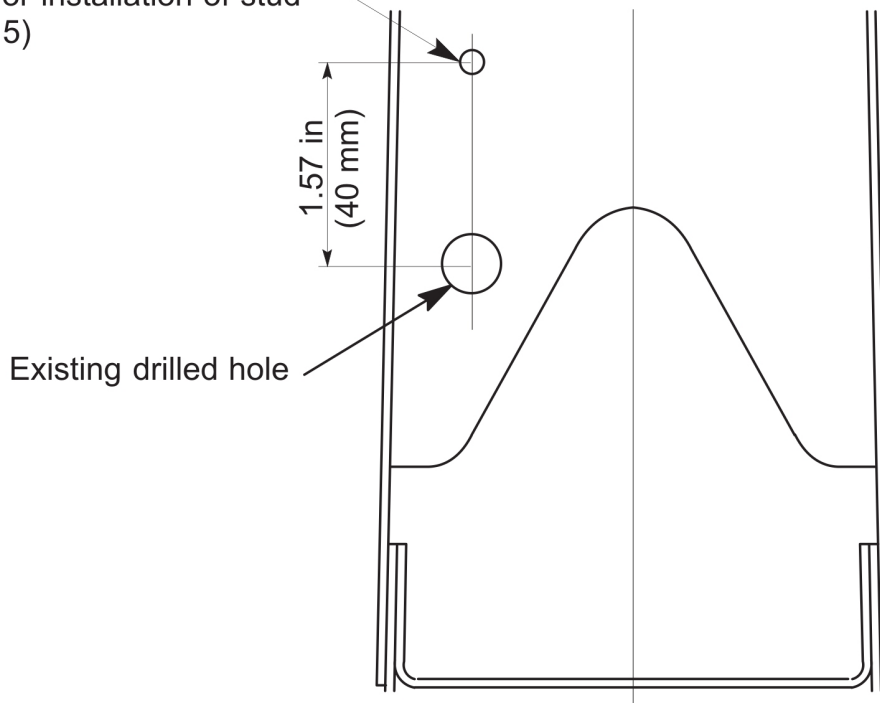


Figure 7: FAIRING P/N F50B319BX OR FGFB319A2 (A3) - RE-ROUTING THE WIRING - AIRCRAFT WITH WIRING ROUTED AROUND THE BACK OF FAIRING

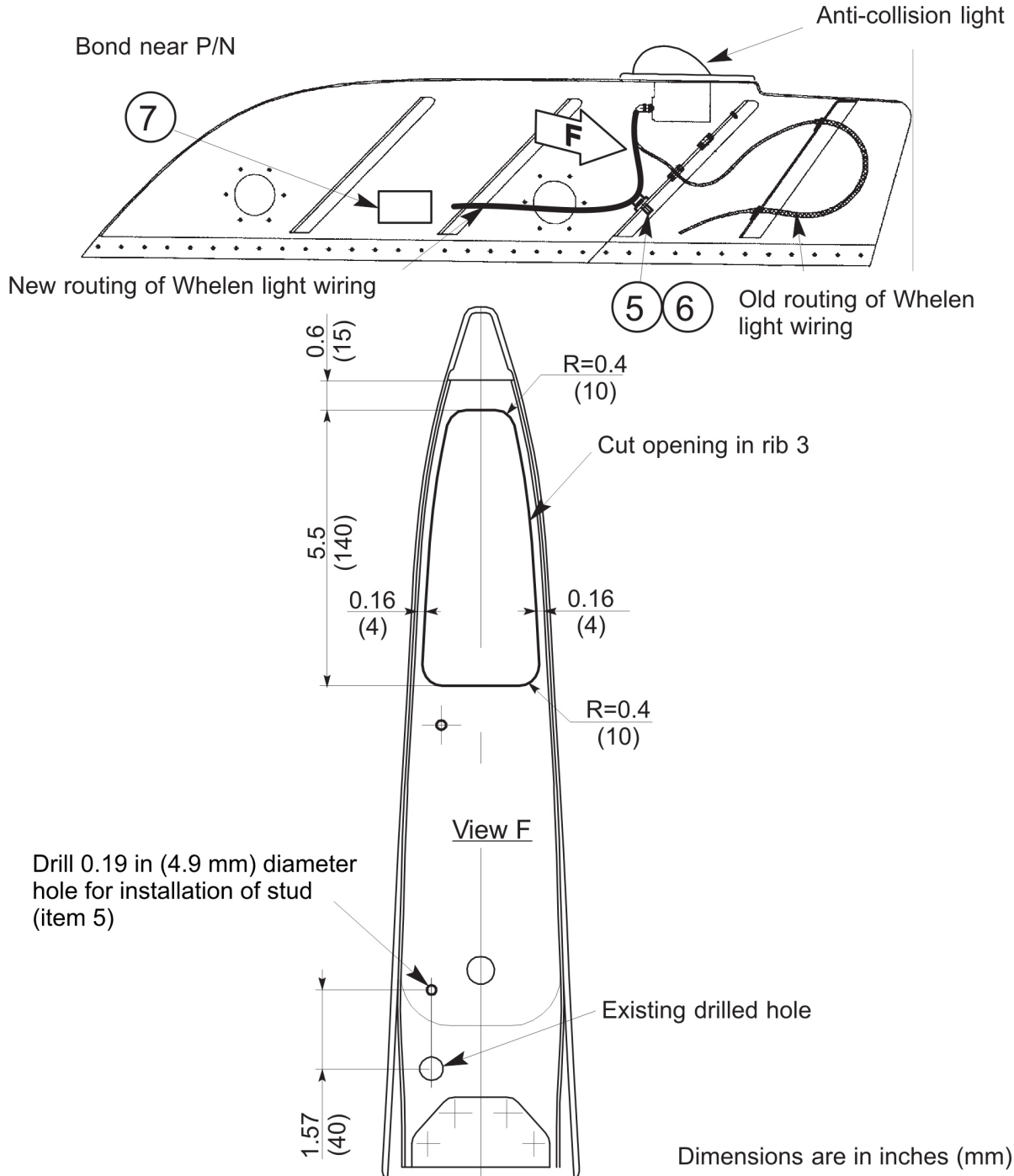


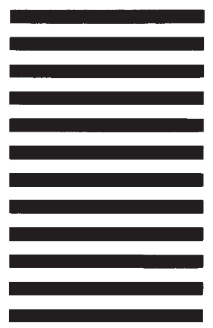
Figure 8: OPENING IN RIB 3 ON A/C EQUIPPED WITH FAIRINGS F50B319, (A1), (A2) AND FGFB319, (A1)

FROM _____

ZIP CODE _____



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IN THE
UNITED STATES



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ATTENTION:
FIELD SERVICE DEPARTMENT



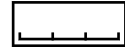
DASSAULT AVIATION
78, Quai Marcel Dassault
Cedex 300
92552 SAINT-CLOUD Cedex
FRANCE

A l'attention de DSC

SERVICE BULLETINS COMPLIANCE

FALCON 900

A/C S/N



DASSAULT AVIATION Number
 F900 372

ATA Number
 F900 33 10

Coding:

- C** complete compliance
- T** partial compliance

Performed by:

Remarks: Revision 1

Date:

Signature:

SERVICE BULLETINS COMPLIANCE

Would you be kind enough to inform us that this Service Bulletin is complied with:

- For "Western Hemisphere", mail upper card (No postage necessary if mailed in the U.S.A.) or fax to Dassault Falcon Jet: 1-201-541-4700.
- For "Other Continents", mail card herebelow or fax to Dassault Aviation: 33 (0)1.47.11.65.21.

On the occasion, if other Service Bulletins have been simultaneously complied with could you complete the blank spaces provided for this purpose.

APPLICATION DE BULLETINS DE SERVICE

Lorsque vous aurez applique ce Bulletin de Service sur votre avion, veuillez avoir l'amabilite de nous en informer:

- Utiliser la carte reponse superieure pour continent americain (Dispensee de timbrage pour postage depuis les U.S.A.) ou le fax pour Dassault Falcon Jet : 1-201-541-4700.
- Ou la carte reponse inferieure pour autres continents ou le fax pour Dassault Aviation : 33 (0)1.47.11.65.21.

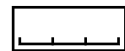
Eventuellement, si d'autres Bulletins de Service ont ete appliques simultanement, vous pouvez les mentionner dans les cases vierges prevues a cet effet.

APPLICATION DES BULLETINS DE SERVICE

SERVICE BULLETINS COMPLIANCE

FALCON 900

A/C S/N



DASSAULT AVIATION Number
 Numero DASSAULT AVIATION
 F900 372

ATA Number
 Numero ATA
 F900 33 10

Coding:

- C** complete compliance
 application totale
- T** partial compliance
 application partielle

Performed by:
 Fait par:

Remarks: Revision 1

Date:

Signature:

Remarques: Revision 1